

THE IDENTIFICATION OF THE INSTABILITY CORE OF INSTABILITIES UNDERPINNED BY AN ACOUSTIC-HYDRODYNAMIC FEEDBACK.

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Acoustic-hydrodynamic feedbacks are a common theme in jet noise. Strong sound emissions are supported by fluid instabilities, whose core is not necessarily localized in space. A common example is the feedback-loop instability of cavity flows, impinging jets or the flow past airfoils. The feedback-loop is composed of a convective instability, which is usually an instability of the shear layer, and an acoustic pressure wave or a hydrodynamic non-local effect. Despite the fact that such a mechanism is widely accepted, a precise identification of the most sensitive spatial regions underpinning the instability is missing. Herein, we propose a non-local decomposition of the *structural sensitivity* [1] and the *endogeneity* [2] concepts. These notions allow us to precisely identify the most sensitive regions of the flow responsible for the closure of the feedback-loop. The systematic use of these techniques could be applied in the design of passive flow control devices.

First, we decompose the direct global mode, respectively the adjoint; the velocity field follows the Helmholtz-Hodge decomposition,

$$\hat{\mathbf{u}} = \hat{\mathbf{u}}_{ac} + \hat{\mathbf{u}}_{hyd} = \nabla \phi_c + \nabla \times \Psi, \quad \hat{\mathbf{u}}^\dagger = \hat{\mathbf{u}}_{hyd}^\dagger + \hat{\mathbf{u}}_{ac}^\dagger = \nabla \phi_c^\dagger + \nabla \times \Psi^\dagger. \quad (1)$$

In a second step, we analyse the effect of a localised harmonic forcing $\mathbf{H}(\hat{\mathbf{q}}) \equiv \delta(\mathbf{x} - \mathbf{x}_0) \mathbf{P}_H \mathbf{C}_0 \mathbf{P}_{\hat{\mathbf{q}}} \hat{\mathbf{q}}$ on

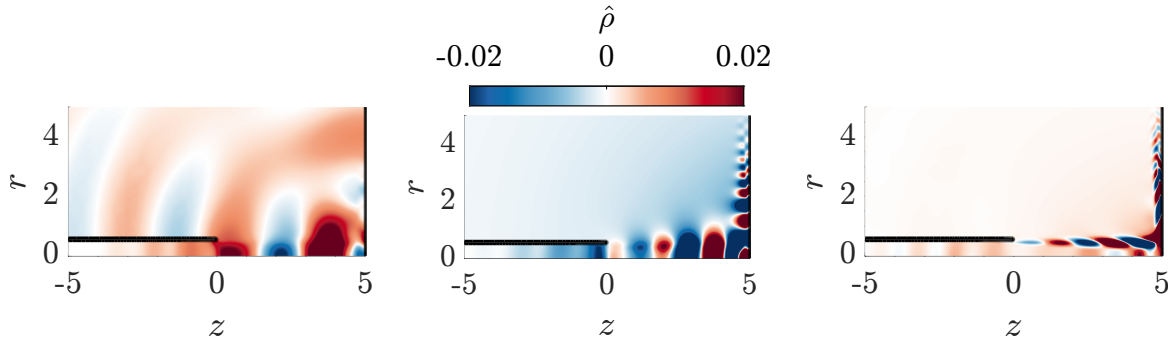


Figure 1: Decomposed density field of a mode ($Re = 900$, $M_J \approx 0.9$). (left) $\hat{\rho}_{ac}$, (centre) $\hat{\rho}_{hyd}$, (right) $\hat{\rho}_s$.

the linearised dynamics $\left(-i\omega \mathbf{B}|_{\mathbf{q}_0} + \mathbf{D}\mathbf{F}|_{\mathbf{q}_0} \right) \hat{\mathbf{q}} = \mathbf{H}(\hat{\mathbf{q}})$ by means of a *decomposed* structural sensitivity

$$\begin{aligned} i\delta\omega_j^k &= \langle \hat{\mathbf{u}}_k^\dagger, \delta(\mathbf{x} - \mathbf{x}_0) \mathbf{C}_0 \hat{\mathbf{u}}_j \rangle \leq \|\mathbf{C}_0\| \|\hat{\mathbf{u}}_k^\dagger(\mathbf{x}_0)\| \|\hat{\mathbf{u}}_j(\mathbf{x}_0)\| = \|\mathbf{C}_0\| \mathbf{S}_{\mathbf{u},s}^{(j,k)}(\mathbf{x}_0), \\ \mathbf{S}_{\mathbf{u},s}^{(j,k)}(\mathbf{x}_0) &= \|\hat{\mathbf{u}}_k^\dagger(\mathbf{x}_0)\| \|\hat{\mathbf{u}}_j(\mathbf{x}_0)\| \text{ with } j, k = ac, hyd, s. \end{aligned} \quad (2)$$

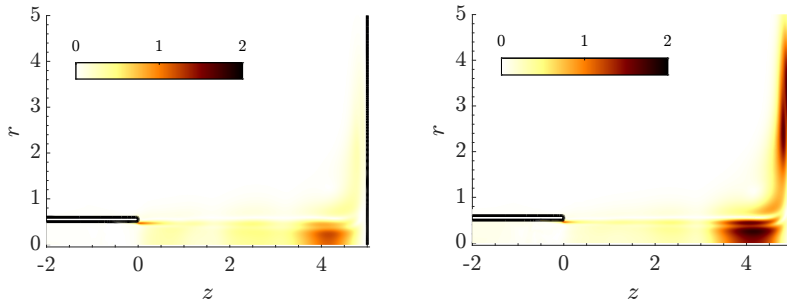


Figure 2: (left) $\mathbf{S}_{\mathbf{u},s}^{(hyd,ac)}$, (right) $\mathbf{S}_{\mathbf{u},s}^{(ac,hyd)}$ for the same global mode as Fig. 1.

References

- [1] Giannetti, Flavio, and P. Luchini (2007). *Structural sensitivity of the first instability of the cylinder wake*. Journal of Fluid Mechanics, 581, 167-197.
- [2] Marquet, Olivier, and L. Lesshafft. *Identifying the active flow regions that drive linear and nonlinear instabilities* arXiv preprint arXiv:1508.07620 (2015).